



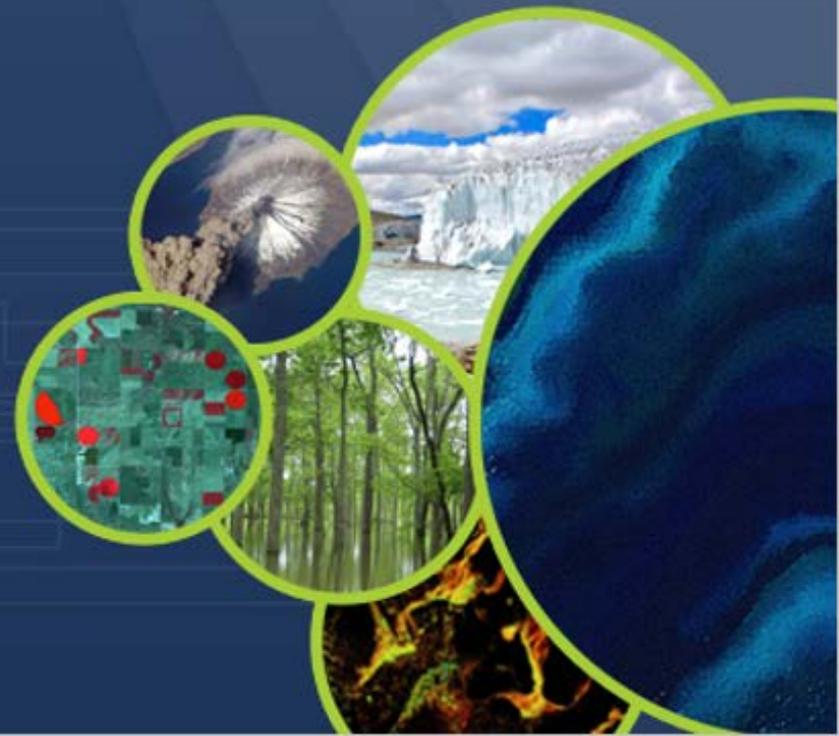
Committee on Earth Observation Satellites

Calibration of GHG Sensors

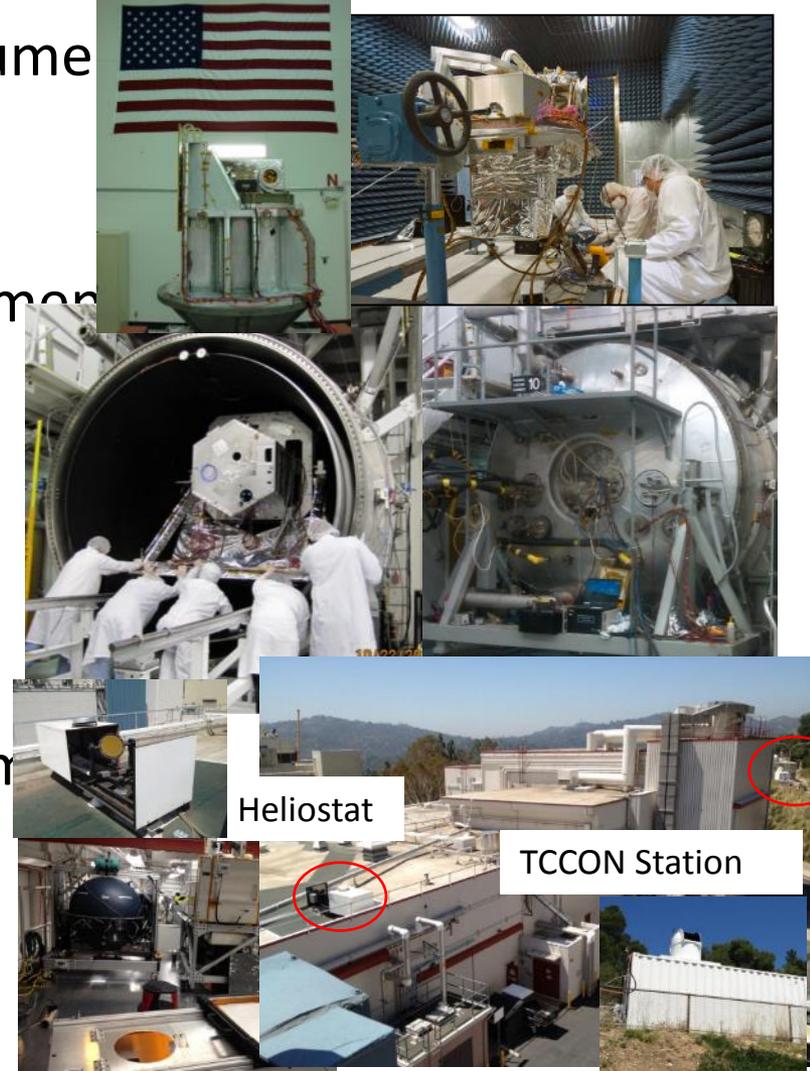
Report to CEOS AC-VC

June 28, 2017

David Crisp (Jet Propulsion Laboratory,
California Institute of Technology)



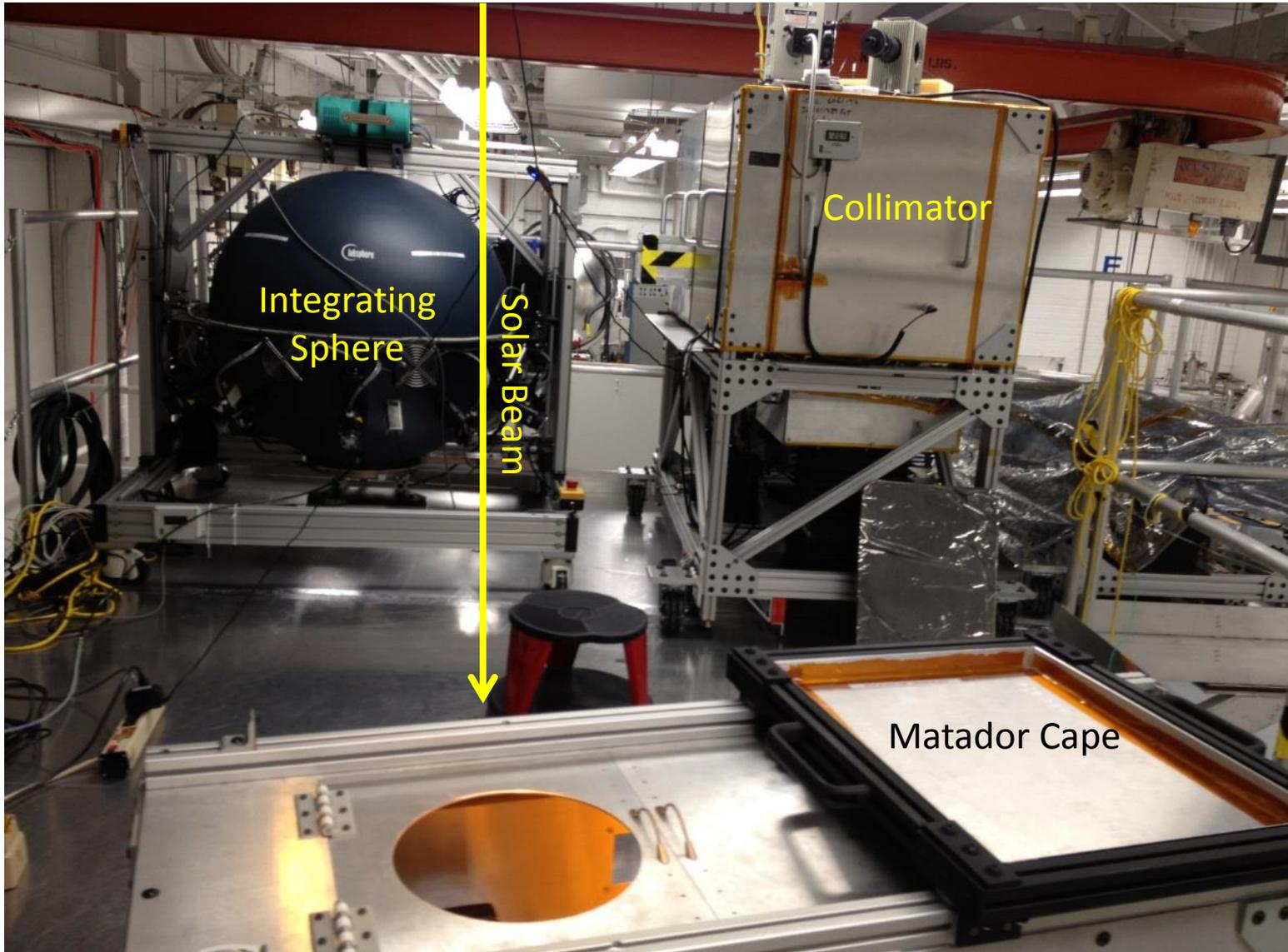
- Pre-flight testing quantifies key Instrument knowledge parameters
 - **Geometric**
 - Field of view, Bore-sight alignment
 - **Radiometric**
 - Zero-level offset (bias)
 - Gain, Gain non-linearity
 - **Spectroscopic**
 - Spectral range, resolution, sampling
 - Instrument Line Shape (ILS)
 - **Polarization**
 - **Instrument stability**





OCO-2 employed four types of optical ground support equipment

- **Collimator:** spatially-defined continuum and laser light sources to
 - Establish the spectrometer focus
 - Define the instrument field of view (including slit alignment, spatial stray light)
 - Define the spectrometer instrument line shape and spectral scattered light
 - Determine the angle of polarization
- **Integrating Sphere:** spatially uniform continuum light sources to
 - Characterize and calibrate radiometric performance (minimum and maximum measurable signal, radiometric gain and its linearity, signal to noise ratio)
 - Provide a baseline for the pixel-to-pixel variability in gain
- **Step-scan FTS:** for assessing spectral stray light rejection
- **Heliostat:** acquire atmospheric spectra using direct sunlight
 - Validate the instrument line shape and dispersion
 - Test instrument linearity and transient response over a range of illumination levels
 - Provide an end-to-end test of instrument calibration & retrieval algorithm performance, through comparisons with TCCON XCO₂ retrievals



Integrating Sphere

Solar Beam

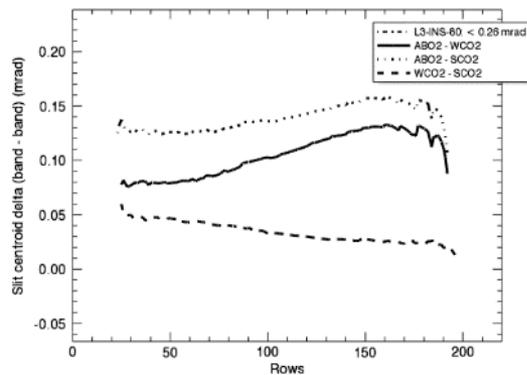
Collimator

Matador Cape

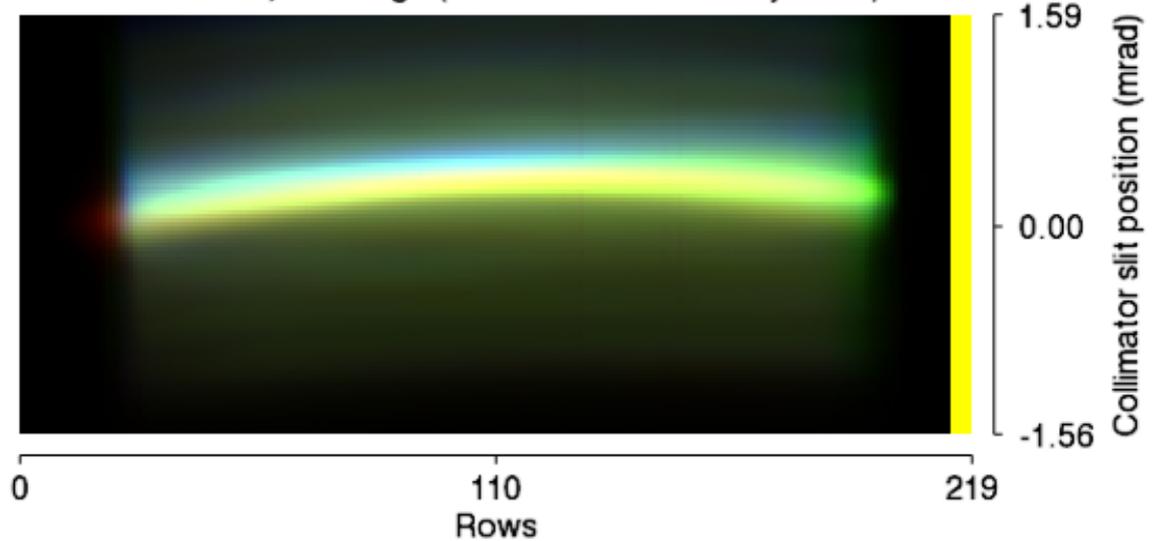


Requirement	Value	Measured	Notes
Slit Width	< 2 mrad	~ 0.5 mrad (typical) 0.7 mrad (worst case)	~3x Margin
Slit Misalignment	< 0.26 mrad	~ 0.1 mrad (typical) 0.15 mrad (worst case)	~70% Margin

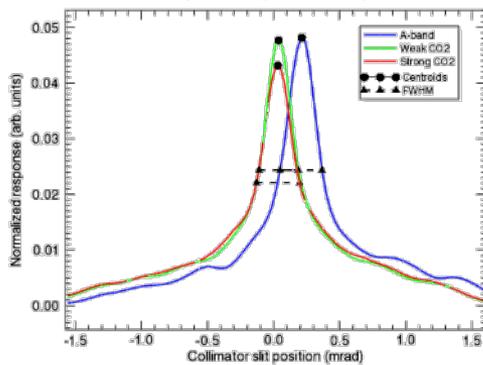
Test: 0260, Relative slit alignment



Test: 0260, Slit image (rows not measured = "yellow")



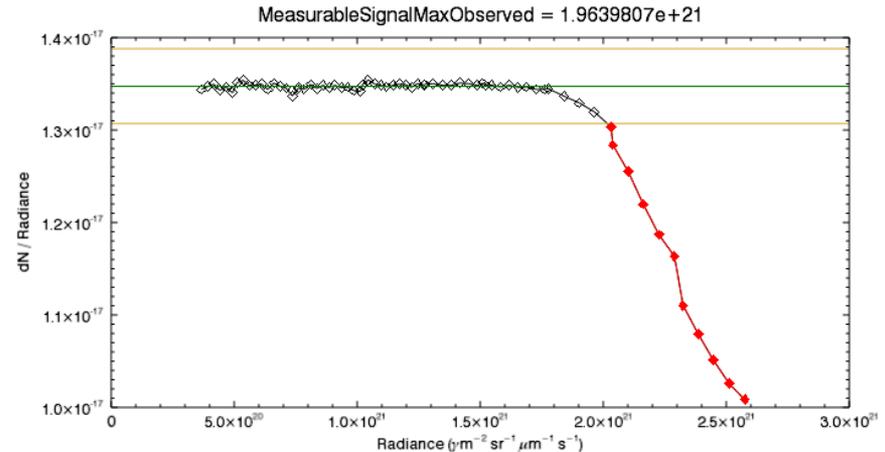
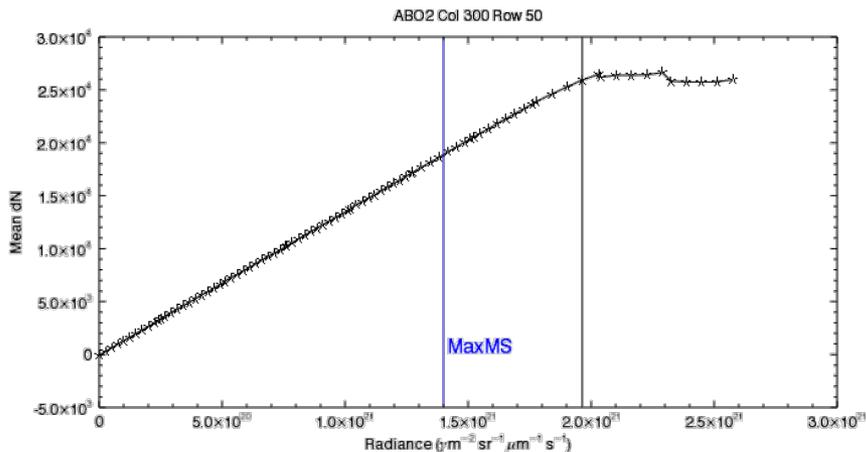
Test: 0253, Color slice (area normalized): row=100



Impacts of slit misalignment and scene non-uniformity mitigated by defocusing the entrance telescope.



Requirement	Value	Measured	Notes
Max Measureable Signal – A-band	$\geq 1.4 \times 10^{21} *$	$\sim 1.8 \times 10^{21} *$	<ul style="list-style-type: none"> ~30% Margin
Max Measureable Signal – Weak CO ₂	$\geq 4.9 \times 10^{20} *$	$> 8.7 \times 10^{21} *$	<ul style="list-style-type: none"> Very large margins Sphere isn't bright enough to saturate the detectors in CO₂ channels
Max Measureable Signal – Strong CO ₂	$\geq 2.5 \times 10^{20} *$	$> 3.8 \times 10^{21} *$	

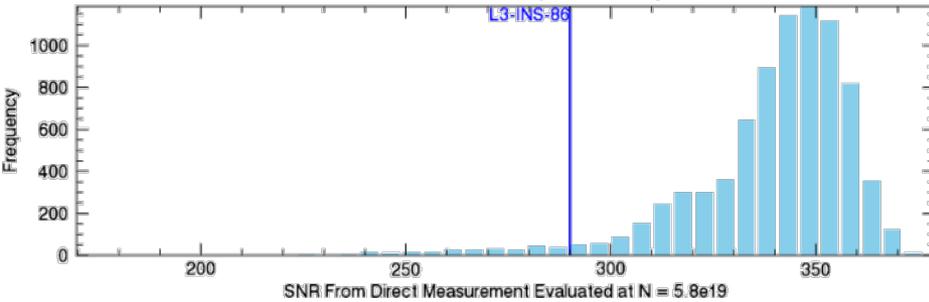


* OCO Radiance Units are: photons/m²/sr/μm/s

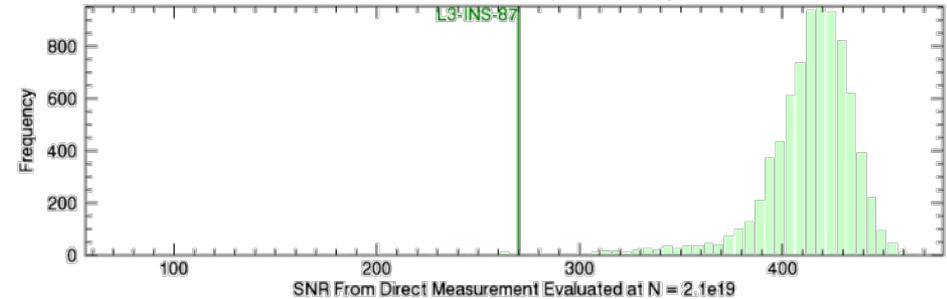


Requirement	Value	Measured	Notes
Signal-to-Noise Ratio – A-band	> 290	302 – 361	At 5.9×10^{19} photons/m ² /sr/μm/s
Signal-to-Noise Ratio – Weak CO ₂	> 270	369 - 441	At 2.1×10^{19} photons/m ² /sr/μm/s
Signal-to-Noise Ratio – Strong CO ₂	≥ 190	267 - 350	At 1.1×10^{19} photons/m ² /sr/μm/s

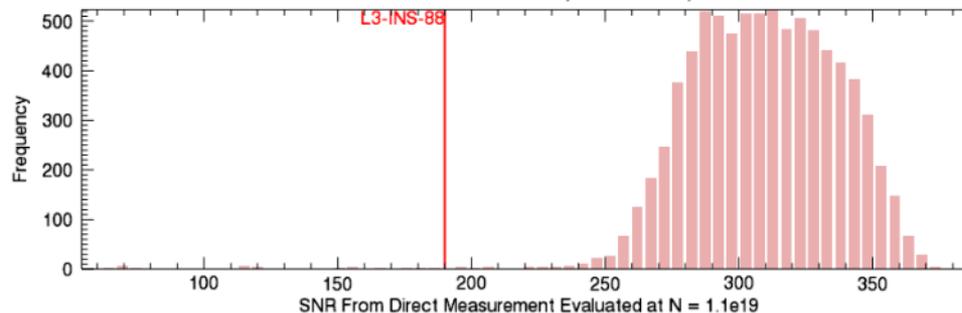
ABO2 Nominal Radiance: 5% = 301.6, 50% = 342.9, 95% = 360.6



WCO2 Nominal Radiance: 5% = 369.0, 50% = 416.6, 95% = 441.0

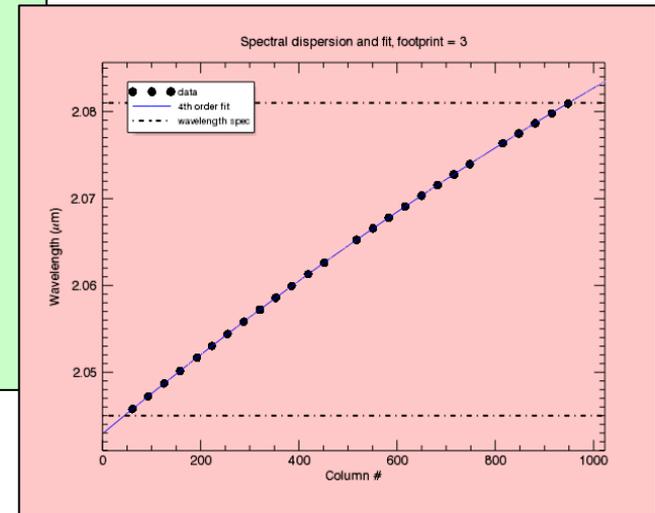
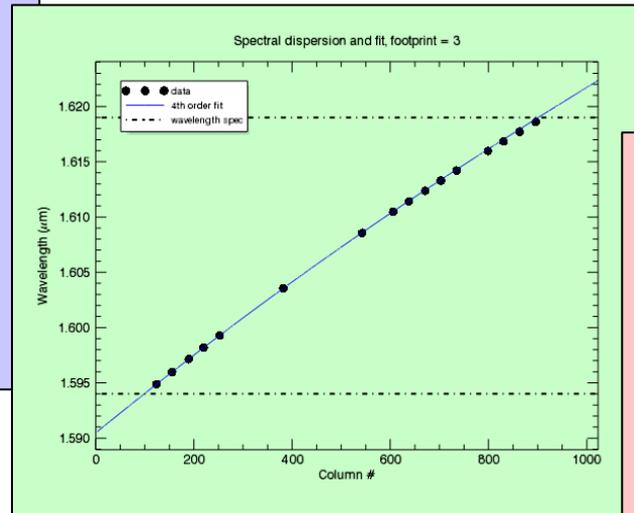
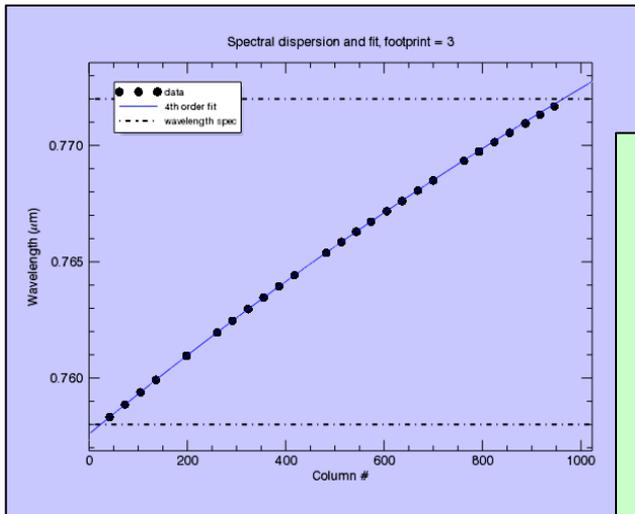


SCO2 Nominal Radiance: 5% = 266.6, 50% = 308.9, 95% = 350.4



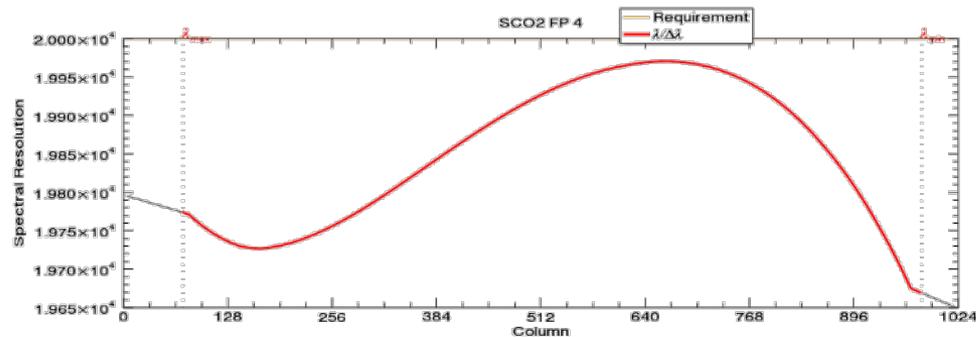
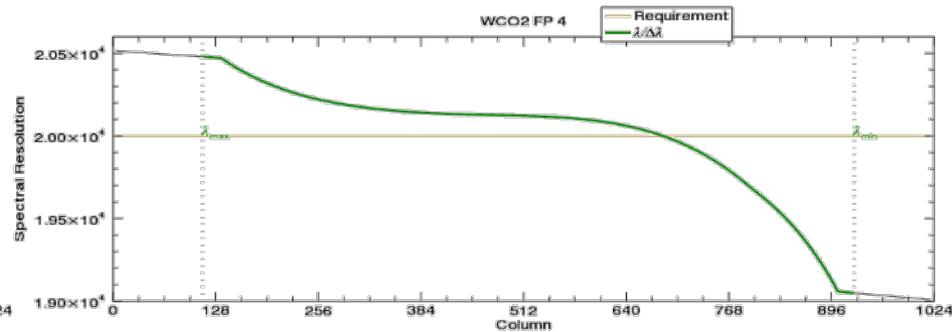
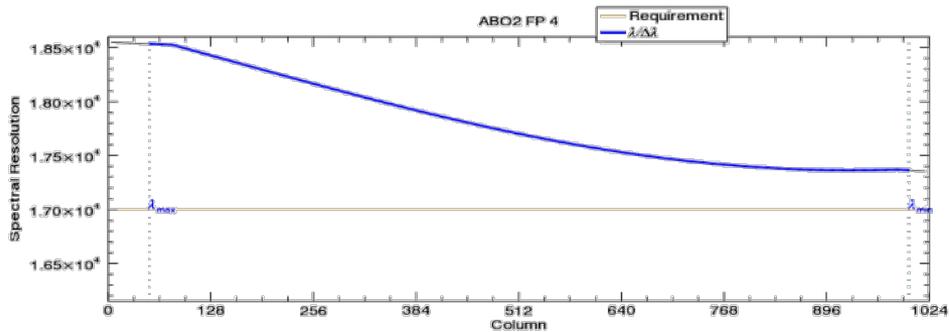


Requirement	Value	Measured	Notes
Spectral Range – A-band	758 to 772 nm	757.6 – 772.6 nm	Bands are well centered for OCO-2
Spectral Range – Weak CO ₂	1,594 – 1,619 nm	1,590.6 – 1,621.8 nm	
Spectral Range – Strong CO ₂	2,045 – 2,082 nm	2,043.1 – 2083.3 nm	



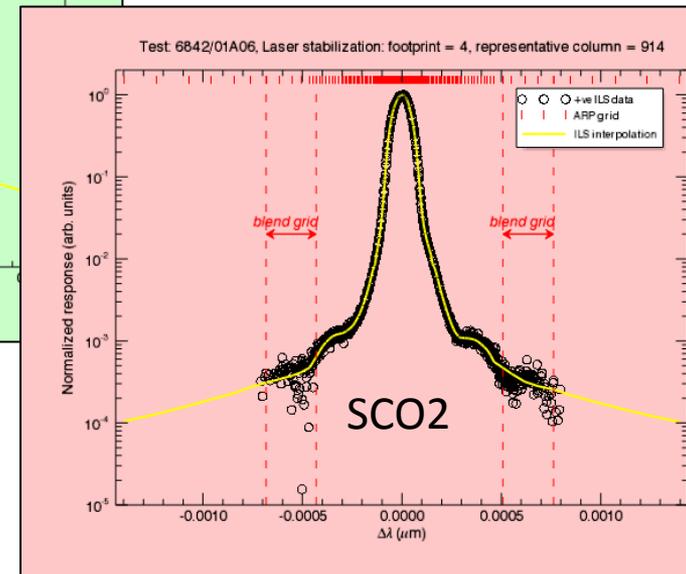
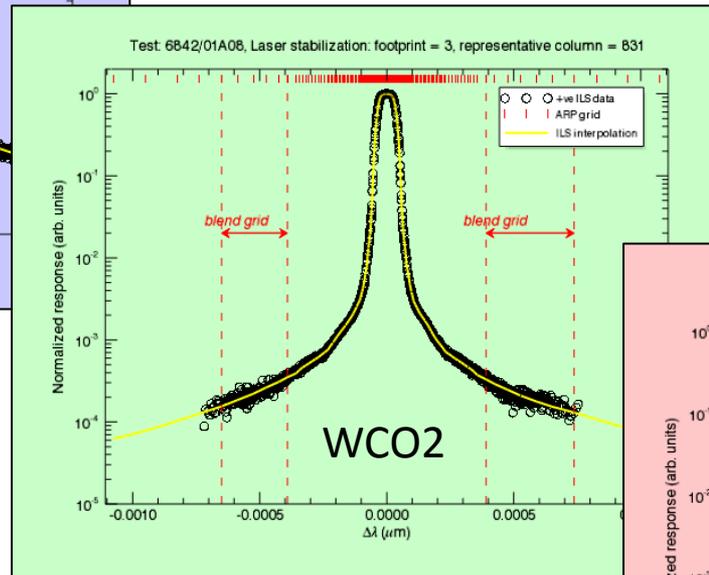
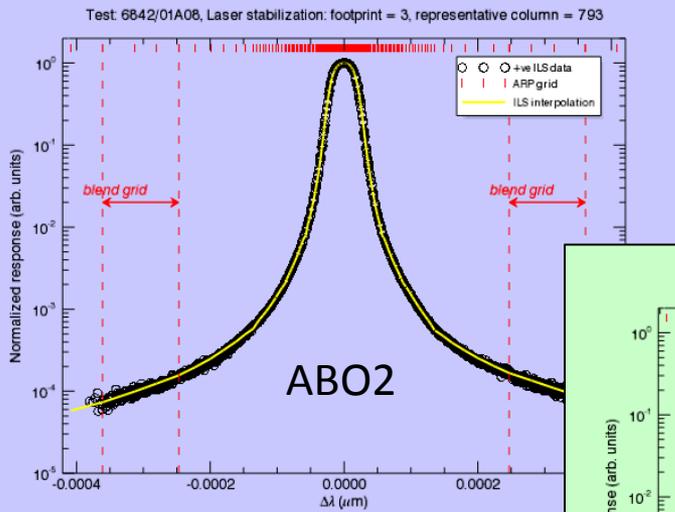


Requirement	Value	Measured	Notes
Spectral Resolution – A-band	> 17,000	17,500 – 18,500	Resolving power is slightly low in CO ₂ channels. L2 Algorithm Team found found no impact on OCO-2 Level 1 requirements
Spectral Resolution – Weak CO ₂	> 20,000	19,100 – 20,500	
Spectral Resolution – Strong CO ₂	> 20,000	19,700 – 19,900	





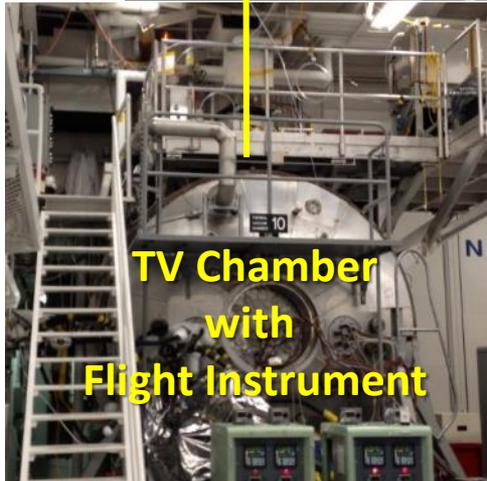
- Tunable diode lasers were used to characterize the width and shape of the ILS across each of the three spectral ranges.
- This method could characterize the ILS shape over a dynamic range of 1000 to 10000.



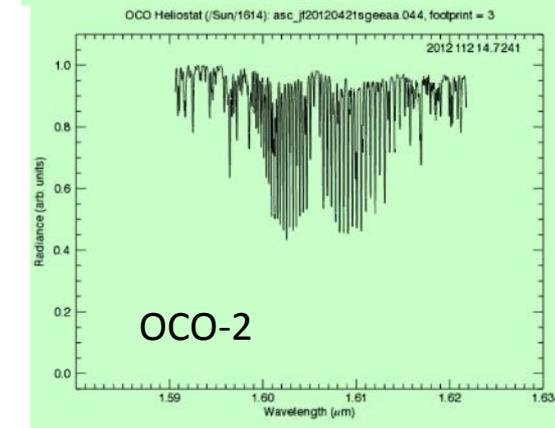
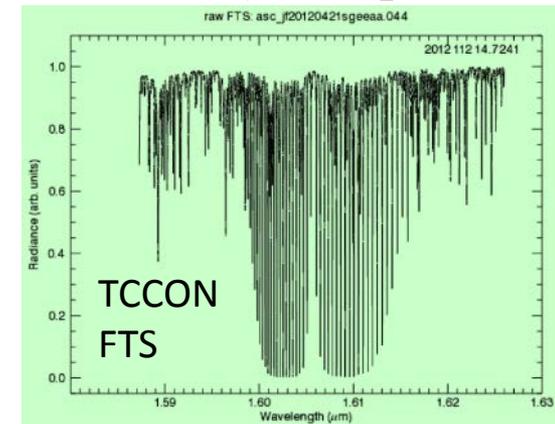
These results provided valuable information about the ILS shape, width (resolving power) and dispersion, but these results were not adequate to meet the OCO-2 requirements



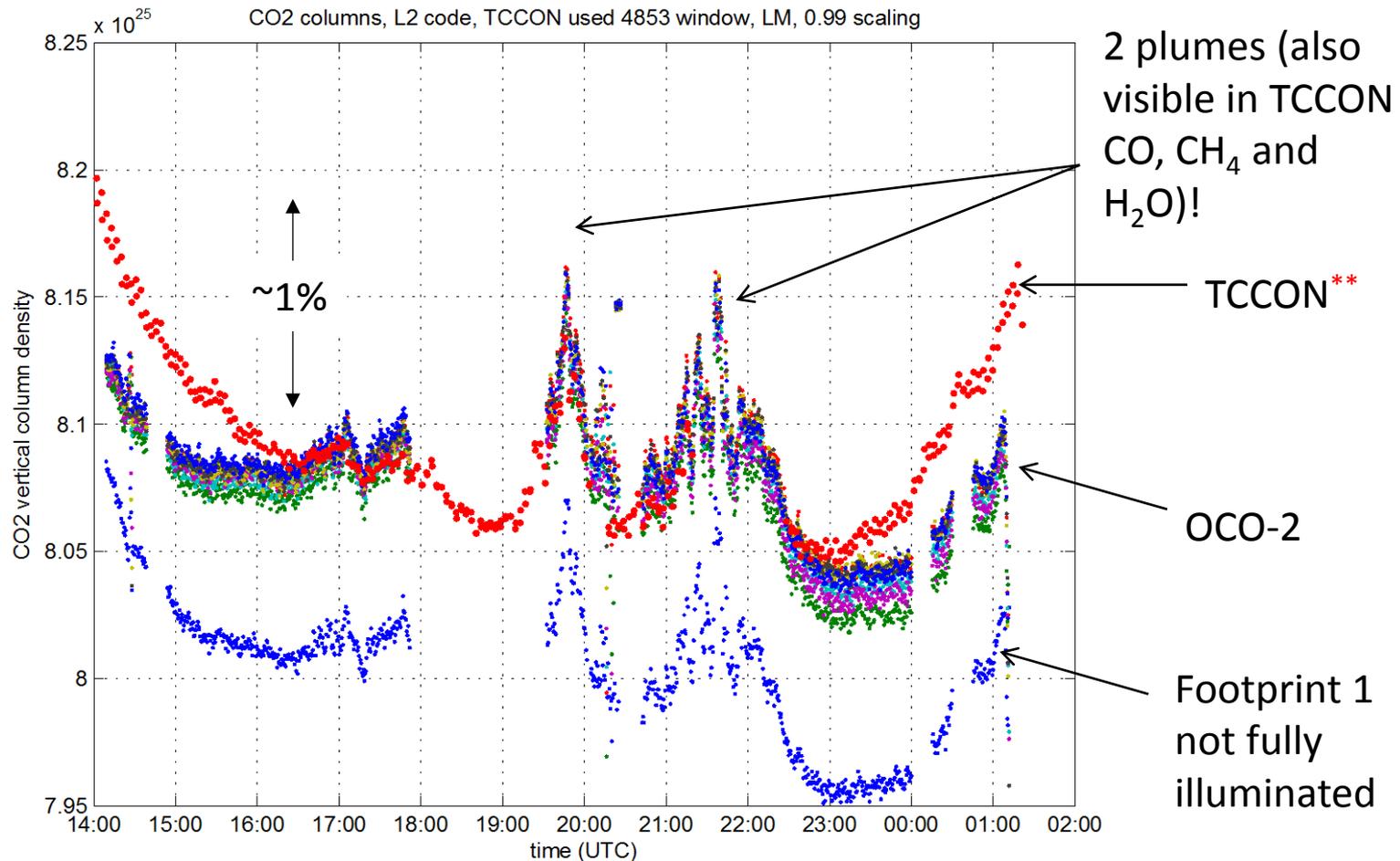
Observations of the sun with the flight instrument taken during TVAC tests provide an end-to-end verification of the instrument performance.



1.6 μm CO₂



21 April 2012

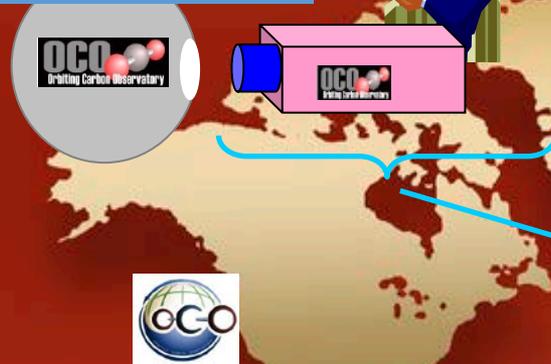


7 of the 8 footprints in the SCO2 channel produce CO₂ column estimates within $\pm 0.25\%$.

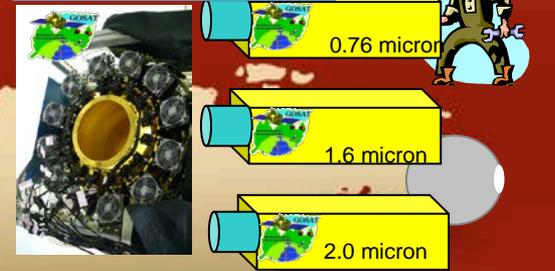
** TCCON does not use this channel to retrieve X_{CO_2} . This is a custom retrieval by D. Wunch.



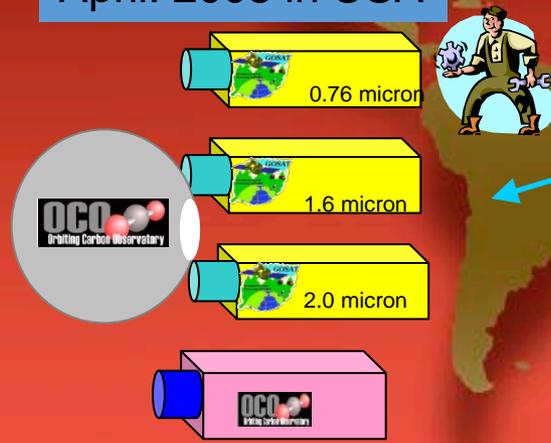
Summer 2007



Summer 2007

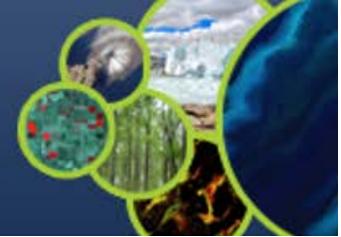


April. 2008 in USA



August 2008 in JAPAN



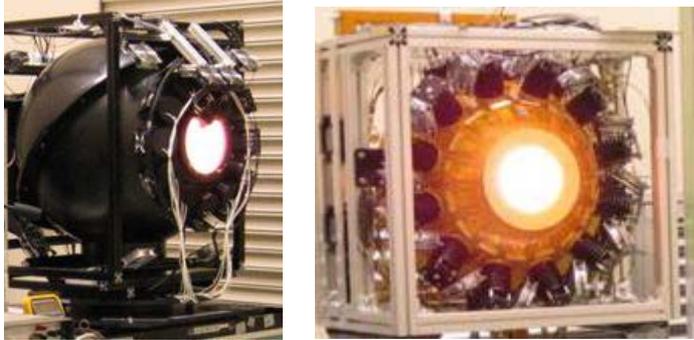


Intercomparison of OCO and GOSAT radiometric standards at JPL in April 2008 and then at Tsukuba in December of 2008

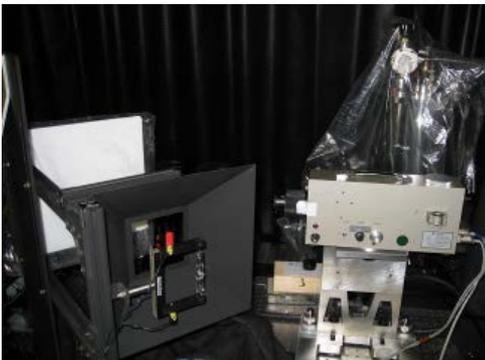




GOSAT Inner-Illuminated Integrating Spheres (1m (BaSO4) and 50 cm (Gold))



NIST standard lamp + Spectralon Diffuser



Portable standard radiometers 3 spectral bands (GOSAT) 3 detectors (OCO)



0.76 micron (B1)



1.6 micron (B2)



2.0 micron (B3)



0.76 micron (AO2)



1.6 micron (WCO2)



2.0 micron (SCO2)

GOSAT ASD (Field Spec)



OCO ASD (Field Spec)





- The OCO and GOSAT radiometric standards showed
 - Excellent agreement in linearity over the full range of illumination conditions considered
 - The O₂ A-band and 1.61 micron CO₂ radiometers showed very good agreement (<1% differences) in radiometric gain
 - The 2-micron radiometers showed larger differences
 - Traced to spatial inhomogeneity within the integrating sphere, due to reduced reflectance of coating
 - Other issues associated with temperature drift in radiometers

The logo for the Committee on Earth Observing Satellites (CEOS). It features the letters 'CEOS' in a bold, green, sans-serif font. The letter 'O' is replaced by a stylized globe of the Earth, showing blue oceans and green continents. The entire logo is enclosed within a white circular border that has a slight 3D effect.

CEOS

Post-Launch Calibration



Credit: Jeff Sullivan

Verifying Radiometric Calibration: The On-board Calibration System



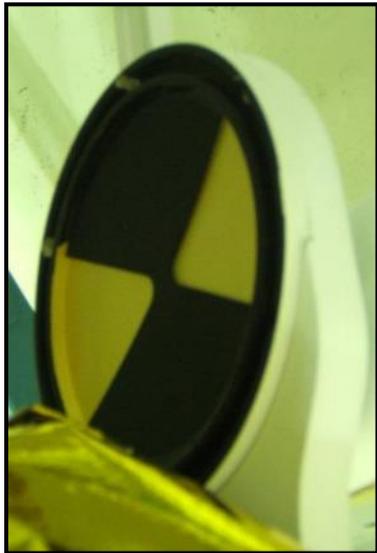
Open for Science observations

Closed for lamp calibration



The on-board calibration (OBC) system consists of a rotating calibration paddle that carries:

- an aperture cover, with a reflective diffuser illuminated by on-board lamps for monitoring pixel-to-pixel variations
- A transmission diffuser for making observations of the solar disk for monitoring radiometric calibration



Reflective diffuser



Telescope baffle assembly, showing lamps for flat fields



O₂ A-Band



Weak CO₂



Strong CO₂

Lamp "flat fields" from each channel.

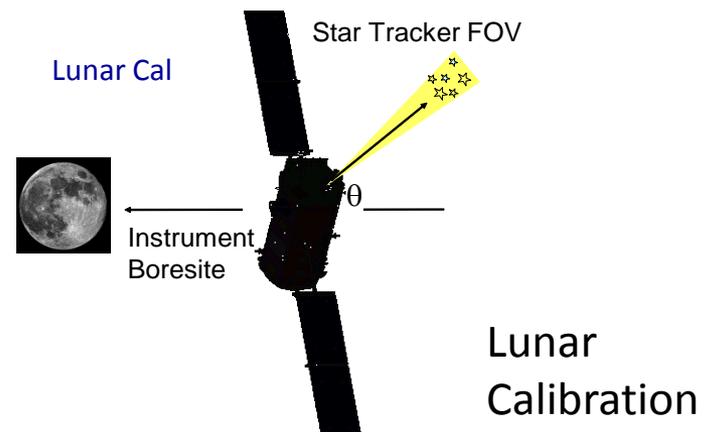
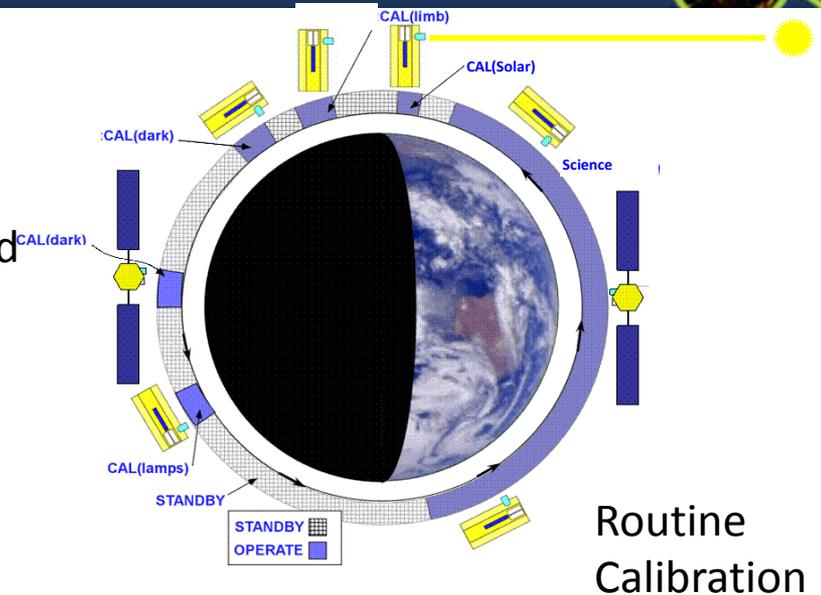


Routine Calibration (every orbit)

- OCO-2 will look at the sun through a solar diffuser
- Dark calibration with aperture door closed and lamps off

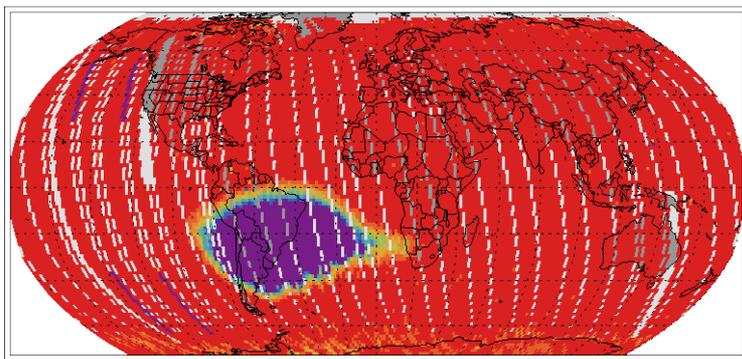
Special Calibration Activities

- Solar Doppler calibration
 - Observe sun through an entire daylight side of an orbit to calibrate ILS
 - once every six months)
- Lunar calibration required for absolute and relative pointing
 - Verifies alignment between instrument bore sight and the star tracker.
 - Used in radiance calibration
 - performed once every lunar month





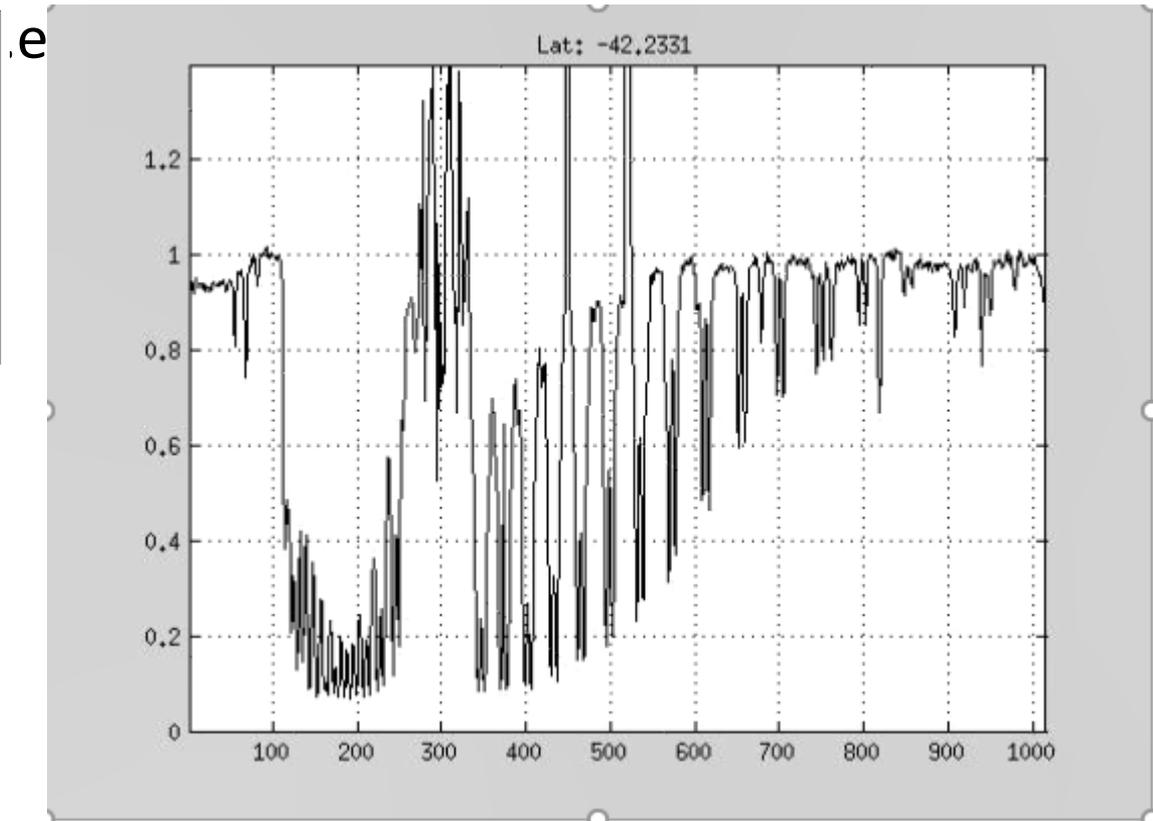
Cosmic rays a particular problem, especially on orbits that pass through the



meas-mod < 10sigma (A-band)

1000 1001 1002 1003 1004 1006 1007 1008 1009 1011 1012 1013 1014 1016

- The largest effects are seen in the O₂ A-band.
- An algorithm to screen the specific colors affected by cosmic rays has been implemented.

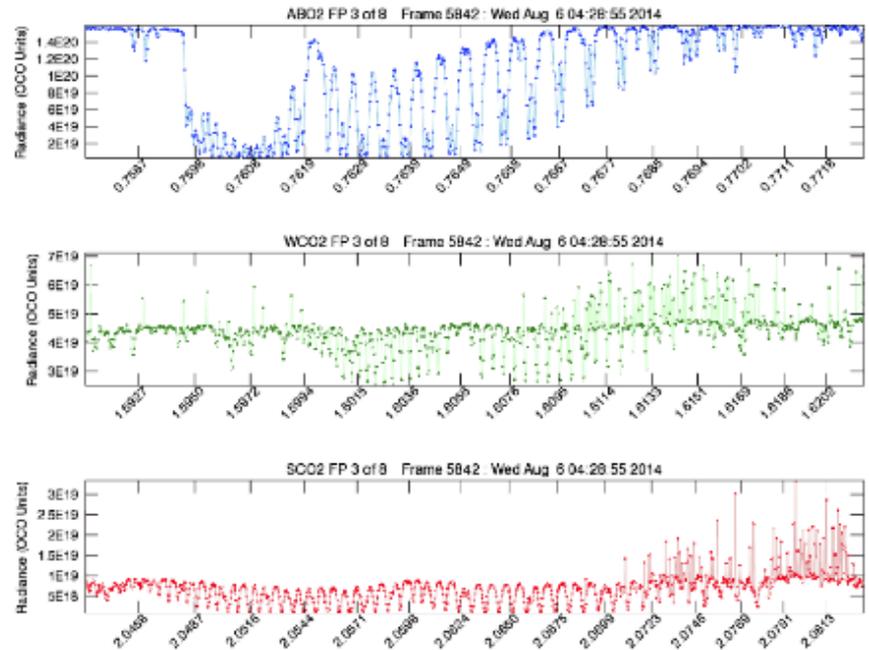
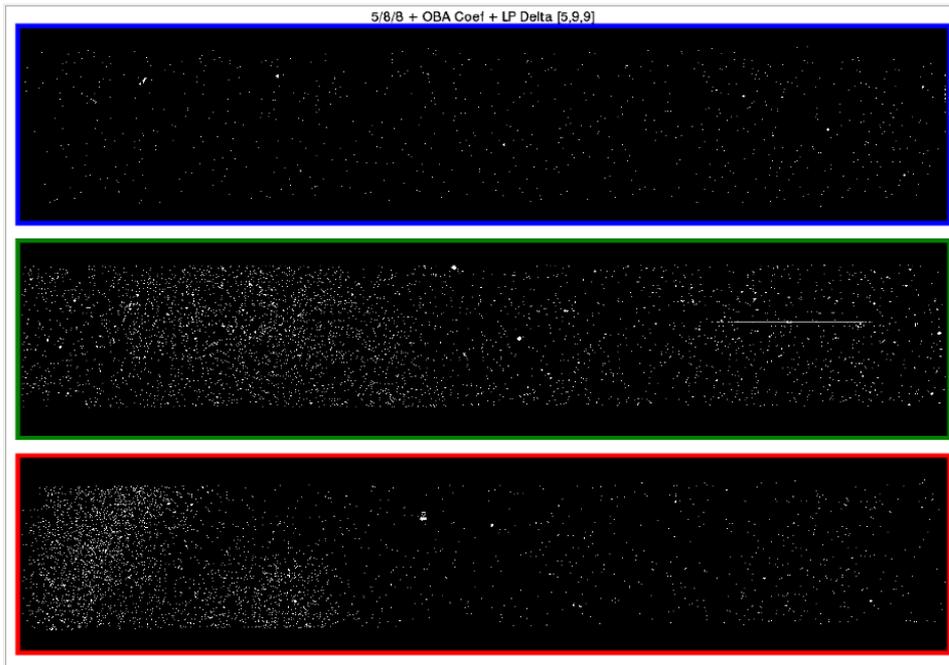


OCO-2 A-band spectra from the South Atlantic Anomaly



The number of bad pixels increased substantially between the pre-flight testing (April 2012) and launch (July 2014)

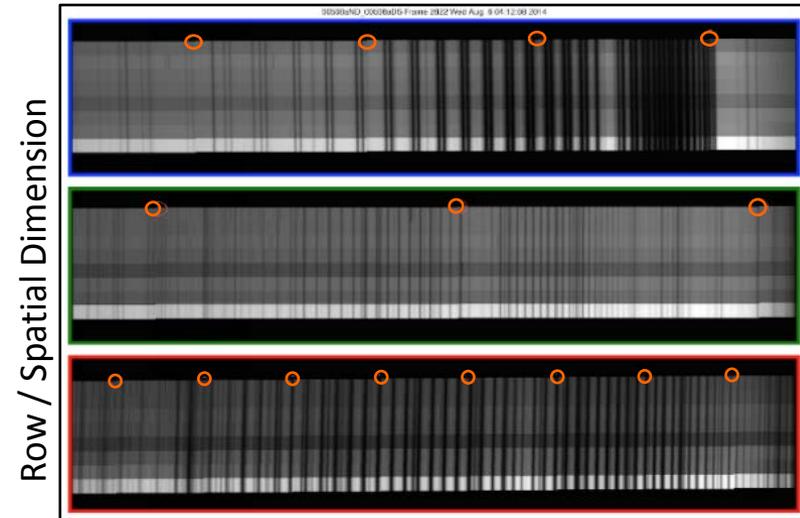
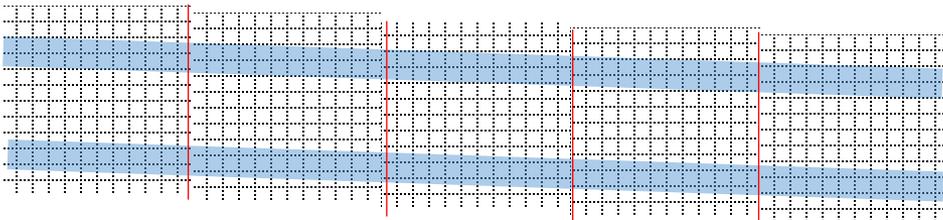
- Improved bad pixel maps were an early focus of Calibration team



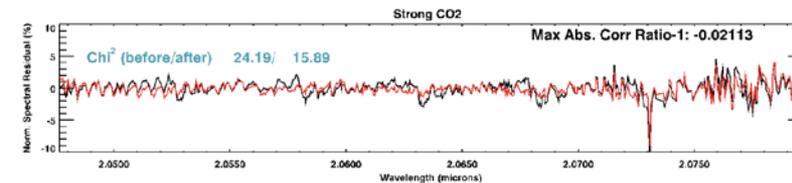
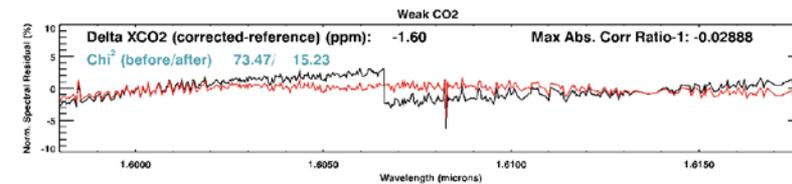
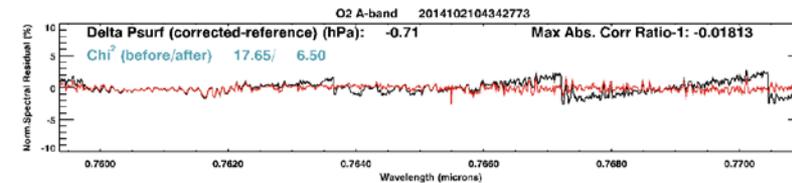
Bad pixels (left) are shown along with their associated bad samples (right) for the A-Band (top), Weak CO₂ (middle) and Strong CO₂ (bottom) channels



- The OCO-2 FPA's are rotated slightly with respect to the slit and grating
- With these *FPA Clocking Errors*, the FPA rows recording a given spatial footprint varies across the spectral range (columns)
- To record the same spatial footprint across an entire spectrum, the starting pixel of each spatial footprint can be adjusted from one column (wavelength) to another (by one pixel)

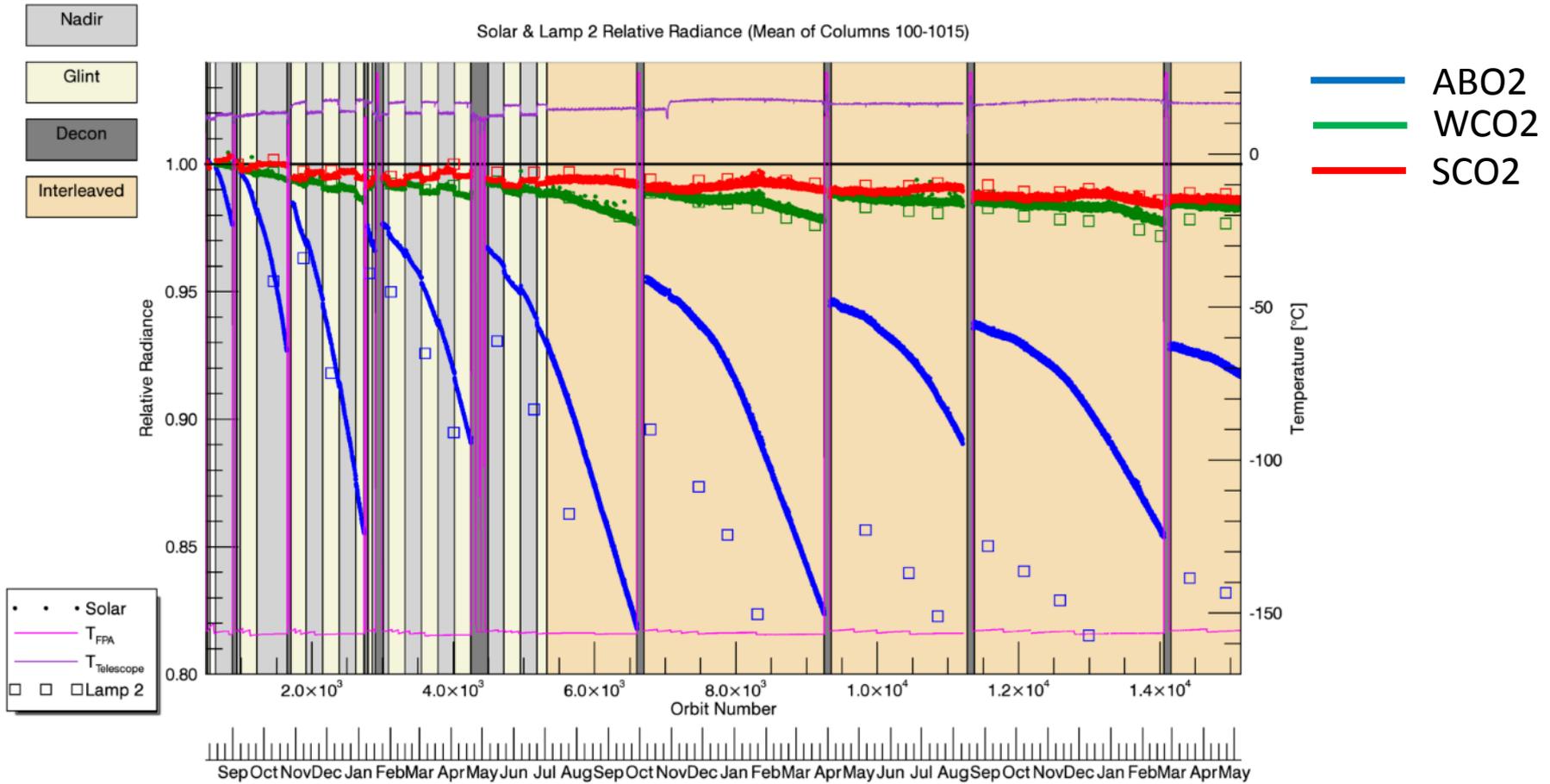


Christian Frankenberg & Fabiano Oyafuso





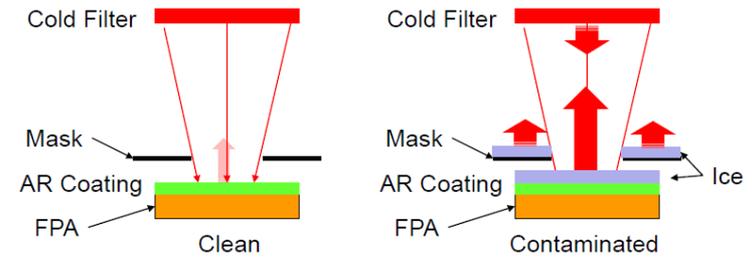
- The sensitivity of the OCO-2 ABO2 channel has varied over time, while the WCO2 and SCO2 show much less variability
- The ABO2 sensitivity degradation has two components
 - A “fast degradation” reversed by decontamination activities
 - This component has been attributed to temporary degradation of the anti-reflection coating on the A-band focal plane array detector (FPA) due to the accumulation of a thin (< 100 nm) layer of ice on the FPA
 - A monotonic “slow degradation”
 - Lunar and Vicarious Calibration measurements indicate that this change is due to degradation of the solar diffuser rather than a throughput loss in the instrument



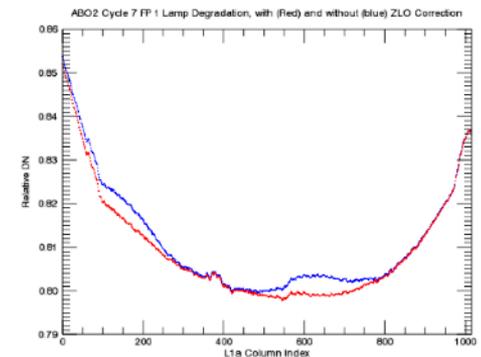
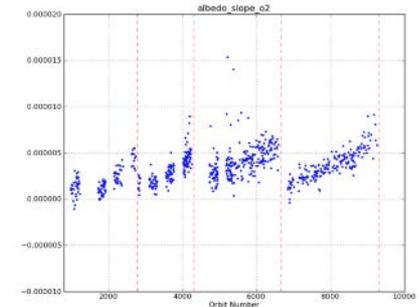
Rate of ice accumulation continues to decrease.

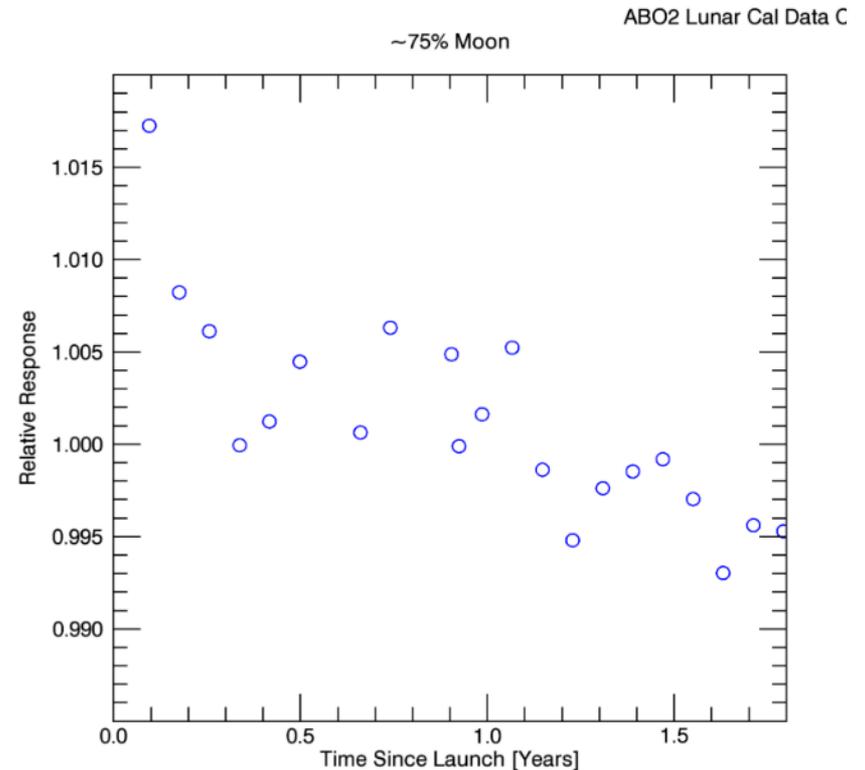
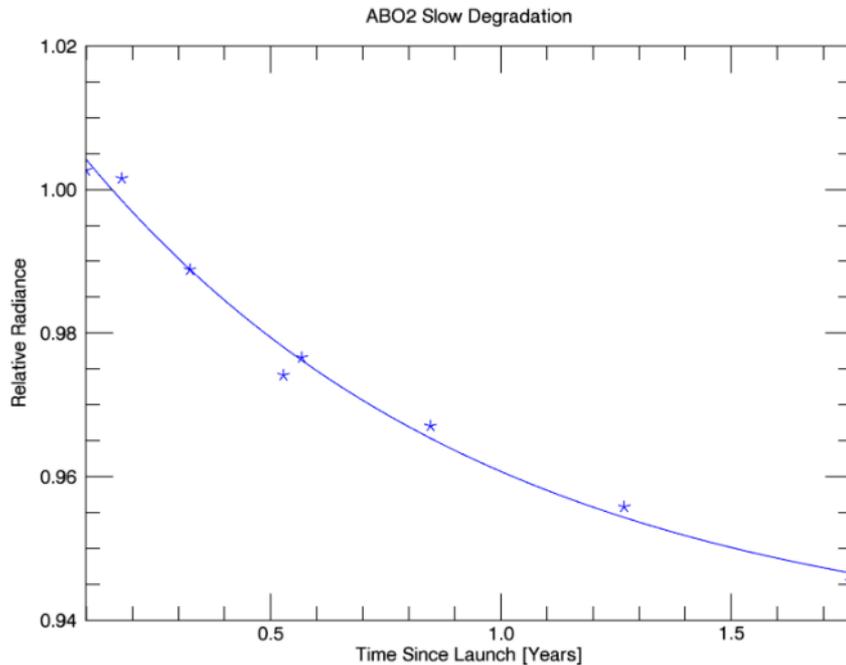


- The “fast degradation is associated with ice accumulation on the A-band (ABO2) focal plane array (FPA), that degrades the performance of the anti-reflection coating
- The ice contamination also introduces a scattered light that introduces a zero level offset (ZLO), which in turn introduces artifacts in the SIF and aerosol retrievals
- Both the reduced signal and the zero level associated with ice accumulation and the ZLO produced by the scattered light are corrected in version 8 L1B radiances



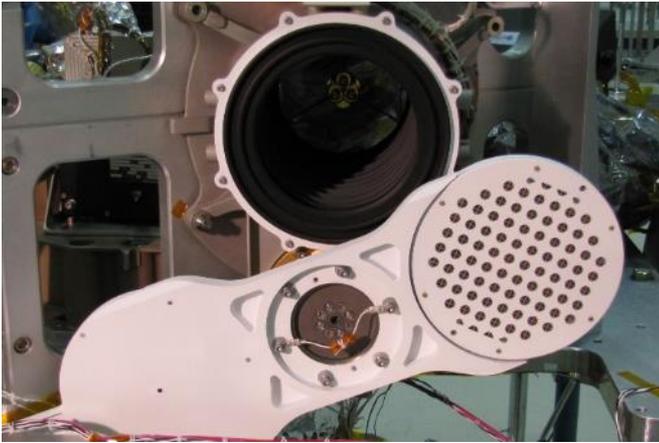
pre-v8 Apply EOF results: O2A Alb Slope



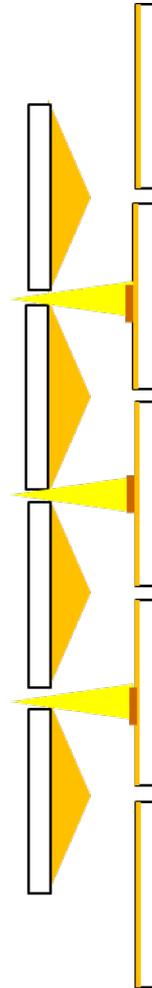


The “slow” component of the degradation is *mostly* due to the calibrator, not the instrument.

A reanalysis of the Lunar Calibration Data indicates that about 20% of the slow degradation is in the instrument optics



The solar diffuser consists of a pair of plates with a series of pin holes that are offset from each other, with a gold coated internal surface

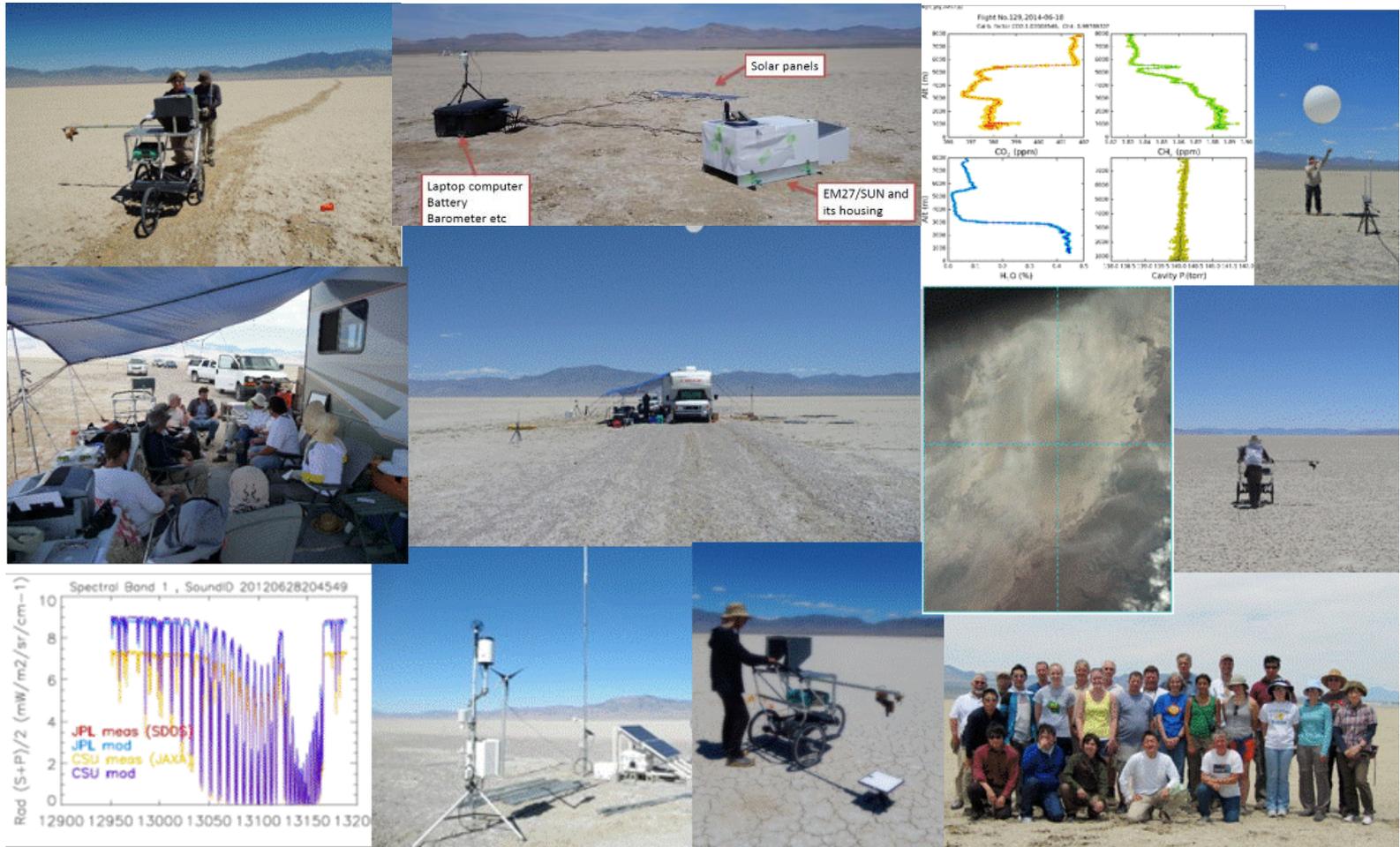


Cartoon showing basic principle only

Hypothesis;

- As solar UV interacts with contaminants on the gold coated inter surface, it causes darkening.
- The most severe darkening is expected in the ABO2 Channel
- The rate of the darkening is uniform, but could not be predicted prior to launch.

Vicarious Calibration in Railroad Valley



Vicarious Calibration Campaigns in Railroad Valley have continued to play a critical role in GOSAT inflight calibration, and now play a similar role for OCO-2.



- The GOSAT and OCO-2 instruments were extensively characterized and calibrated prior to launch
- As always happens, the instrument you test on the ground is not the one that arrives in space
 - Significant increase in bad pixels in CO₂ bands
 - Sensitivity to non-uniformly illuminated footprints
- Occasionally, you miss things in ground testing
 - Performance reserve is a major contributor to mission success
- Characterizing and correcting for these issues on orbit has been a challenge, but has been enabled by a robust calibration program
- Future missions will face many of these challenges, as well as others associated with:
 - The 2.3 micron channel, which is not shared by OCO-2
 - Those specific to instrument design and vantage point